

Gravitational Fields	Name:	
Long Answer Questions	Class:	
	Date:	

Time:	286 minutes
Marks:	249 marks
Comments:	

Both gravitational and electric field strengths can be described by similar equations written in the form

1

$$a = \frac{bc}{d^2}.$$

(a) Complete the following table by writing down the names of the corresponding quantities, together with their SI units, for the two types of field.

symbol	-	onal field		cal field
	quantity	SI unit	quantity	SI unit
а	gravitational field strength			
b			$\frac{1}{4\pi\varepsilon_0}$	m F ^{−1}
С				
d				

- (b) Two isolated charged objects, A and B, are arranged so that the gravitational force between them is equal and opposite to the electric force between them.
 - (i) The separation of A and B is doubled without changing their charges or masses. State and explain the effect, if any, that this will have on the resultant force between them.
 - (ii) At the original separation, the mass of A is doubled, whilst the charge on A and the mass of B remain as they were initially. What would have to happen to the charge on B to keep the resultant force zero?

(3) (Total 7 marks)

(4)

(a) State Newton's law of gravitation.

2

(b)	two	798 Cavendish investigated Newton's law by measuring the gravitational force between unequal uniform lead spheres. The radius of the larger sphere was 100 mm and that the smaller sphere was 25 mm.
	(i)	The mass of the smaller sphere was 0.74 kg. Show that the mass of the larger sphere was about 47 kg.

density of lead = 11.3 × 10^3 kg m⁻³

(ii) Calculate the gravitational force between the spheres when their surfaces were in contact.

answer = _____ N

(2)

(2)

(c)	between them, were considered in order to improve the accuracy of Cavendish's experiment. Describe and explain the effect on the calculations in part (b) of dou	
	radius of both spheres.	
		(4) (Total 10 marka)
		(Total 10 marks)
(a	Explain why astronauts in an orbiting space vehicle experience the sensation of weightlessness.	

3

(b) A space vehicle has a mass of 16 800 kg and is in orbit 900 km above the surface of the Earth.

mass of the Earth = 5.97×10^{24} kg radius of the Earth = 6.38×10^{6} m

(i) Show that the orbital speed of the vehicle is approximately 7400 m s⁻¹.

(ii) The space vehicle moves from the orbit 900 km above the Earth's surface to an orbit 400 km above the Earth's surface where the orbital speed is 7700 m s $^{-1}$.

Calculate the total change that occurs in the energy of the space vehicle. Assume that the vehicle remains outside the atmosphere after the change of orbit. Use the value of 7400 m s⁻¹ for the speed in the initial orbit.

change in energy _____ J

(4) (Total 10 marks)

4

(a)

State, in words, Newton's law of gravitation.

The E	Earth's orbit is of mean radius 1.50 $ imes$ 10 11 m and the Earth's year is 365 days I
(i)	The mean radius of the orbit of Mercury is 5.79×10^{10} m. Calculate the lengt Mercury's year.

- **5** The planet Venus may be considered to be a sphere of uniform density 5.24×10^3 kg m⁻³. The gravitational field strength at the surface of Venus is 8.87 N kg⁻¹.

Neptune orbits the Sun once every 165 Earth years.

(ii)

(a) (i) Show that the gravitational field strength g_s at the surface of a planet is related to the the density ρ and the radius *R* of the planet by the expression

$$g_s = \frac{4}{3}\pi GR\rho$$

where G is the gravitational constant.

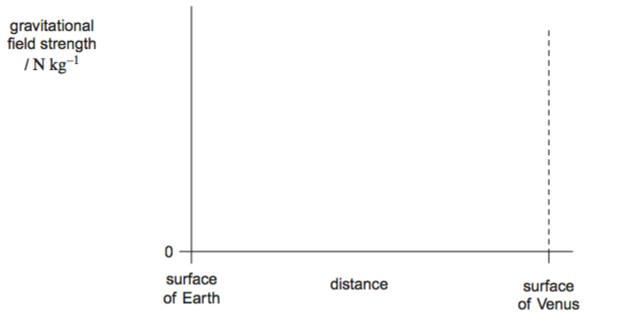
(ii) Calculate the radius of Venus.

Give your answer to an appropriate number of significant figures.

radius = _____ m

(3)

(b) At a certain time, the positions of Earth and Venus are aligned so that the distance between them is a minimum.
 Sketch a graph on the axes below to show how the magnitude of the gravitational field strength *g* varies with distance along the shortest straight line between their surfaces. Consider only the contributions to the field produced by Earth and Venus. Mark values on the vertical axis of your graph.



(3) (Total 8 marks)

The weight *w* of an object on the Earth can be represented either as w = mg or $w = \frac{GMm}{r^2}$. (a) (i) Explain the meaning of *g* and *G* in these equations. Use the equations above to show that $M = \frac{gr^2}{G}$. (ii)

6

(1)

(3)

(iii) Calculate the mass of the Earth to a precision consistent with the data below.

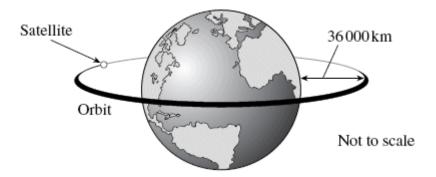
mean radius of the Earth, = 6.4×10^6 m

 $G = 6.7 \times 10^{-11} \text{ N m}^2 \text{ kg}^{-2}$

<i>g</i> =	9.8	Ν	kg ⁻¹
------------	-----	---	------------------

9	5	
	mass of the Earth	kg
		ĸy

(b) The figure below shows a satellite in a geostationary orbit around the Earth.



(i) State the time period for a geostationary satellite.

(1)

(3)

	radius	
Calculate the speed, in km	s ⁻¹ , of a satellite in a geostation	ary orbit.
	speed	km s ⁻¹
State a common use for a g	eostationary satellite.	
Explain why a geostationary	/ orbit is necessary for this use.	



The Rosetta space mission placed a robotic probe on Comet 67P in 2014.

(a) The total mass of the Rosetta spacecraft was 3050 kg. This included the robotic probe of mass 108 kg and 1720 kg of propellant. The propellant was used for changing velocity while travelling in deep space where the gravitational field strength is negligible.

Calculate the change in gravitational potential energy of the Rosetta spacecraft from launch until it was in deep space.

Give your answer to an appropriate number of significant figures.

Mass of the Earth = 6.0×10^{24} kg Radius of the Earth = 6400 km

change in gravitational potential energy ______ J

(4)

(b) As it approached the comet, the speed of the Rosetta spacecraft was reduced to match that of the comet. This was done in stages using four 'thrusters'. These were fired simultaneously in the same direction.

Explain how the propellant produces the thrust.

(c) Each thruster provided a constant thrust of 11 N.

Calculate the deceleration of the Rosetta spacecraft produced by the four thrusters when its mass was 1400 kg.

decleration _____ m s^{-2}

(1)

(d) Calculate the maximum change in speed that could be produced using the 1720 kg of propellants.

Assume that the speed of the exhaust gases produced by the propellant was 1200 m s^{-1}

maximum change in speed _____ m $\rm s^{-1}$

(3)

- (e) When the robotic probe landed, it had to be anchored to the comet due to the low gravitational force. Comet 67P has a mass of about 1.1 × 10¹³ kg. A possible landing site was about 2.0 km from the centre of mass.
 - (i) Calculate the gravitational force acting on the robotic probe when at a distance of 2.0 km from the centre of mass of the comet.

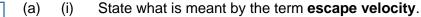
gravitational force _____ N

- (3)
- (ii) Calculate the escape velocity for an object 2.0 km from the centre of mass of the comet.

escape velocity _____ m s⁻¹

(iii) A scientist suggests using a drill to make a vertical hole in a rock on the surface of the comet. The anchoring would be removed from the robotic probe before the drill was used. The drill would exert a force of 25 N for 4.8 s.

Explain, with the aid of a calculation, whether this process would cause the robotic probe to escape from the comet.



(1)

(2)

(ii) Show that the escape velocity, v, at the Earth's surface is given by $v = \sqrt{\frac{2GM}{R}}$

where M is the mass of the Earth and R is the radius of the Earth.

(iii) The escape velocity at the Moon's surface is 2.37×10^{-3} m s⁻¹ and the radius of the Moon is 1.74×10^{6} m.

Determine the mean density of the Moon.

mean density _____ kg m⁻³

(2)

(b) State **two** reasons why rockets launched from the Earth's surface do **not** need to achieve escape velocity to reach their orbit.

(2) (Total 7 marks)



9

(2)

(2)

(ii) A mass m is at a height h above the surface of a planet of mass M and radius R. The gravitational field strength at height h is g. By considering the gravitational force acting on mass m, derive an equation from Newton's law of gravitation to express g in terms of M, R, h and the gravitational constant G.

(b) (i) A satellite of mass 2520 kg is at a height of 1.39×10^7 m above the surface of the Earth. Calculate the gravitational force of the Earth attracting the satellite. Give your answer to an appropriate number of significant figures.

force attracting satellite _____ N

(3)

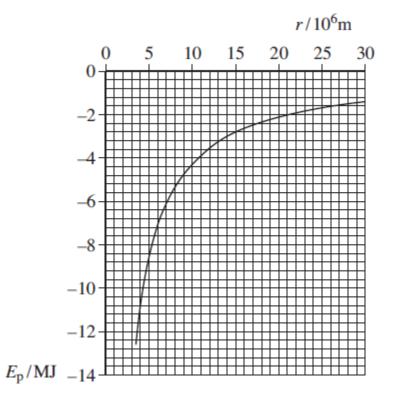
(ii) The satellite in part (i) is in a circular polar orbit. Show that the satellite would travel around the Earth three times every 24 hours.

(5)

(c) State and explain **one** possible use for the satellite travelling in the orbit in part (ii).

(Total 14 marks)

10 The graph below shows how the gravitational potential energy, E_p , of a 1.0 kg mass varies with distance, *r*, from the centre of Mars. The graph is plotted for positions above the surface of Mars.



(a) Explain why the values of E_p are negative.

(b) Use data from the graph to determine the mass of Mars.

(3)
(c) Calculate the escape velocity for an object on the surface of Mars.
escape velocity _____ m s⁻¹
(3)

(d) Show that the graph data agree with $E_{
m p} \propto rac{1}{r}$

(3) (Total 11 marks)

(3)

(b) The table gives the gravitational potentials, *V*, at three different distances, *r*, from the centre of the Earth.

distance from centre of Earth r / km	gravitational potential V / 10 ⁷ J kg ⁻¹
7500	-5.36
12500	-3.22
22500	-1.79

(i) Explain why the gravitational potential at a point in a gravitational field is negative.

(ii)	Show that the data in the table are consistent with	Vα	r-	-1.
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A . 1114 . C	450 her is second form on orbits for dive 7500 her second the Ford
A satellite of mass	450 kg is moved from an orbit of radius 7500 km around the Eart s 12 500 km.
to an orbit of radiu Use data from the	
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to an orbit of radiu Use data from the	s 12 500 km.
to an orbit of radiu Use data from the	s 12 500 km.

- (c) The kinetic energy of a 450 kg satellite orbiting the Earth with a radius of 7500 km is 12 GJ.
 - (i) Calculate the kinetic energy of the 450 kg satellite when it is in an orbit of radius 12 500 km.

mass of the Earth = 6.0×10^{24} kg		
kinetic energy	GJ	
	nto the higher	
Calculate the change in kinetic energy of the satellite when it moves i orbit.	nto the higher	
orbit.		
orbit.	GJ	
change in kinetic energy Calculate the total energy that has to be supplied to move the 450 k	GJ	
change in kinetic energy Calculate the total energy that has to be supplied to move the 450 k	g satellite from	

- **12** The Hubble space telescope was launched in 1990 into a circular orbit near to the Earth. It travels around the Earth once every 97 minutes.
 - (a) Calculate the angular speed of the Hubble telescope, stating an appropriate unit.

answer = _____ (3) (b) (i) Calculate the radius of the orbit of the Hubble telescope. answer = _____ m (3) The mass of the Hubble telescope is 1.1×10^4 kg. Calculate the magnitude of the (ii) centripetal force that acts on it. answer = _____ Ν (2) (Total 8 marks) (a) (i) State the relationship between the gravitational potential energy, E_{p} , and the gravitational potential, V, for a body of mass m placed in a gravitational field.

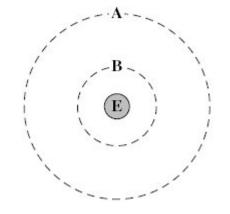
13

(1)

(ii) What is the effect, if any, on the values of E_p and V if the mass m is doubled?

value of <i>E</i> _p	 	
value of V	 	

(b)



The diagram above shows two of the orbits, **A** and **B**, that could be occupied by a satellite in circular orbit around the Earth, **E**.

The gravitational potential due to the Earth of each of these orbits is:

orbit **A** $- 12.0 \text{ MJ kg}^{-1}$ orbit **B** $- 36.0 \text{ MJ kg}^{-1}$.

(i) Calculate the radius, from the centre of the Earth, of orbit **A**.

answer = _____ m

(2)

(2)

(ii) Show that the radius of orbit **B** is approximately 1.1×10^4 km.

(1)

(iii) Calculate the centripetal acceleration of a satellite in orbit **B**.

answer = _____ m s⁻²

(2)

(1)

(iv) Show that the gravitational potential energy of a 330 kg satellite decreases by about 8 GJ when it moves from orbit **A** to orbit **B**.

(c) Explain why it is not possible to use the equation $\Delta E_p = mg\Delta h$ when determining the change in the gravitational potential energy of a satellite as it moves between these orbits.

(1) (Total 10 marks) (a) Explain what is meant by the gravitational potential at a point in a gravitational field.

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(b) Use the following data to calculate the gravitational potential at the surface of the Moon.

mass of Earth $= 81 \times \text{mass}$ of Moon radius of Earth $= 3.7 \times$ radius of Moon gravitational potential at surface of the Earth = -63 MJ kg^{-1} Sketch a graph on the axes below to indicate how the gravitational potential varies with distance along a line outwards from the surface of the Earth to the surface of the Moon.

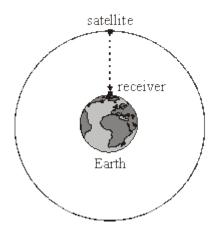
(C)



(3)



The Global Positioning System (GPS) is a system of satellites that transmit radio signals which can be used to locate the position of a receiver anywhere on Earth.

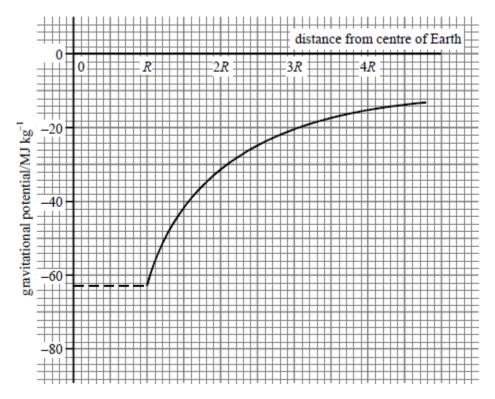


- (a) A receiver at sea level detects a signal from a satellite in a circular orbit when it is passing directly overhead as shown in the diagram above.
 - (i) The microwave signal is received 68 ms after it was transmitted from the satellite. Calculate the height of the satellite.
 - Show that the gravitational field strength of the Earth at the position of the satellite is 0.56 N kg⁻¹.

mass of the Earth	=	6.0 × 10 ²⁴ kg
mean radius of the Earth	=	6400 km

(b)	Fort	the satellite in this orbit, calculate	
	(i)	its speed,	
	(ii)	its time period.	
	()		
		(To	otal 9 m
Com	nmunic	(To cations satellites are usually placed in a <i>geo-synchronous orbit</i> .	otal 9 m
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	State	cations satellites are usually placed in a <i>geo-synchronous orbit</i> . e two features of a geo-synchronous orbit.	
(a)	State	cations satellites are usually placed in a <i>geo-synchronous orbit</i> .	
(a)	State	cations satellites are usually placed in a <i>geo-synchronous orbit</i> . e two features of a geo-synchronous orbit.	
(a)	State	cations satellites are usually placed in a <i>geo-synchronous orbit</i> . e two features of a geo-synchronous orbit. on that the mass of the Earth is 6.00×10^{24} kg and its mean radius is 6.40×10^{6} m show that the radius of a geo-synchronous orbit must be 4.23×10^{7} m,	
(a)	State	e two features of a geo-synchronous orbit.	

- (a) The graph shows how the gravitational potential varies with distance in the region above the surface of the Earth. R is the radius of the Earth, which is 6400 km. At the surface of the Earth, the gravitational potential is -62.5 MJ kg⁻¹.



Use the graph to calculate

- (i) the gravitational potential at a distance 2R from the centre of the Earth,
- (ii) the increase in the potential energy of a 1200 kg satellite when it is raised from the surface of the Earth into a circular orbit of radius 3R.

(b) (i) Write down an equation which relates gravitational field strength and gravitational potential. (ii) By use of the graph in part (a), calculate the gravitational field strength at a distance 2R from the centre of the Earth. Show that your result for part (b)(ii) is consistent with the fact that the surface (iii) gravitational field strength is about 10 N kg⁻¹. (5) (Total 9 marks) State, in words, Newton's law of gravitation. (a) 18

(b) Some of the earliest attempts to determine the gravitational constant, G, were regarded as experiments to "weigh" the Earth. By considering the gravitational force acting on a mass at the surface of the Earth, regarded as a sphere of radius R, show that the mass of the Earth is given by

$$M = \frac{gR^2}{G}$$
,

where g is the value of the gravitational field strength at the Earth's surface.

(c) In the following calculation use these data.

radius of the Moon	= 1.74 × 10 ⁶ m
gravitational field strength at Moon's surface	= 1.62 N kg ⁻¹
mass of the Earth M	$= 6.00 \times 10^{24} \text{ kg}$
gravitational constant G	$= 6.67 \times 10^{-11} \text{ N m}^2 \text{ kg}^{-2}$

Calculate the mass of the Moon and express its mass as a percentage of the mass of the Earth.

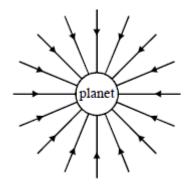
(3) (Total 7 marks)

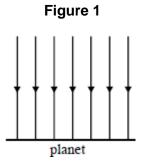
The gravitational field associated with a planet is radial, as shown in **Figure 1**, but near the surface it is effectively uniform, as shown in **Figure 2**.

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20

Alongside each figure, sketch a graph to show how the gravitational potential V associated with the planet varies with distance r (measured outwards from the surface of the planet) in each of these cases.







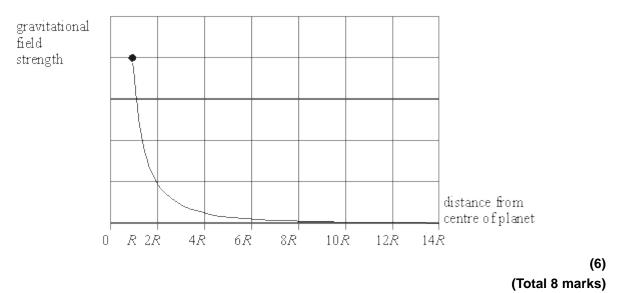
(Total 4 marks)

(a) (i) Explain what is meant by the *gravitational field strength* at a point in a gravitational field.

(ii) State the SI unit of gravitational field strength.

- (b) Planet **P** has mass *M* and radius *R*. Planet **Q** has a radius 3*R*. The values of the gravitational field strengths at the surfaces of **P** and **Q** are the same.
 - (i) Determine the mass of **Q** in terms of *M*.

(ii) The figure below shows how the gravitational field strength above the surface of planet P varies with distance from its centre. Draw on the diagram the variation of the gravitational field strength above the surface of Q over the range shown.

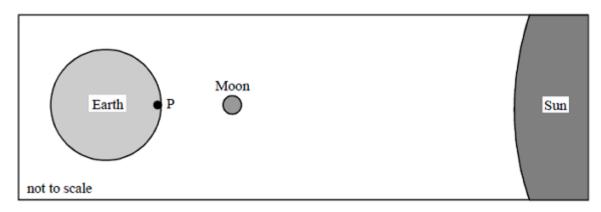


(a) Define *gravitational field strength* at a point in a gravitational field.

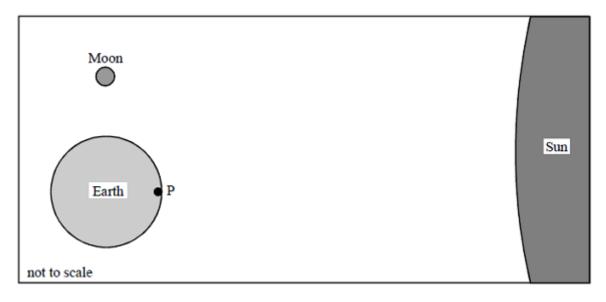
21

(1)

(b) Tides vary in height with the relative positions of the Earth, the Sun and the moon which change as the Earth and the Moon move in their orbits. Two possible configurations are shown in **Figure 1**.



Configuration A



Configuration B

Figure 1

Consider a 1 kg mass of sea water at position **P**. This mass experiences forces $F_{\rm E}$, $F_{\rm M}$ and $F_{\rm S}$ due to its position in the gravitational fields of the Earth, the Moon and the Sun respectively.

(i) Draw labelled arrows on **both** diagrams in **Figure 1** to indicate the three forces experienced by the mass of sea water at **P**.

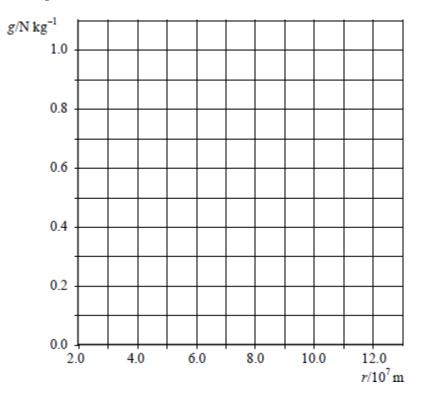
(3)

		(ii)	State and explain which configuration produce the higher tide at position P .	, A or B , of the Sun, the Moon and the Earth will	
					2)
	(c)		ulate the magnitude of the gravitationa n's surface at P , due to the Sun 's gravit	I force experienced by 1 kg of sea water on the actional field.	
			radius of the Earth's orbit	= 1.5 × 10 ¹¹ m	
			mass of the Sun	$= 2.0 \times 10^{30} \text{ kg}$	
			universal gravitational constant,	$G = 6.7 \times 10^{-11} \text{ Nm}^2 \text{ kg}^{-2}$	
				3) Total 9 marks)	-
22	(a)	(i)	Explain what is meant by gravitationa	l field strength.	
		<i>/</i>		(1)
		(ii)	Describe how you would measure the of the Earth. Draw a diagram of the a	e gravitational field strength close to the surface pparatus that you would use.	
				16	5)

(b) (i) The Earth's gravitational field strength (g) at a distance (r) of 2.0 × 10⁷ m from its centre is 1.0 N kg⁻¹. Complete the table with the 3 further values of g.

g/N kg ^{−1}	1.0			
<i>r</i> /10 ⁷ m	2.0	4.0	6.0	8.0

(ii) Below is a grid marked with g and r values on its axes. Draw a graph showing the variation of g with r for values of r between 2.0×10^7 m and 10.0×10^7 m.



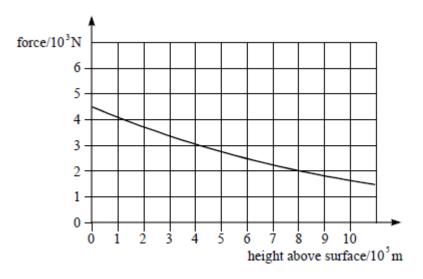
(iii) Estimate the energy required to raise a satellite of mass 800 kg from an orbit of radius 4.0×10^7 m to one of radius 10.0×10^7 m.

(3) (Total 14 marks)

(2)

A lunar landing module and its parent craft are orbiting the Moon at a height above the surface of 6.0×10^5 m. The mass of the lunar module is 2.7×10^3 kg. The graph below shows the variation of the gravitational force on the module with height above the surface of the Moon.

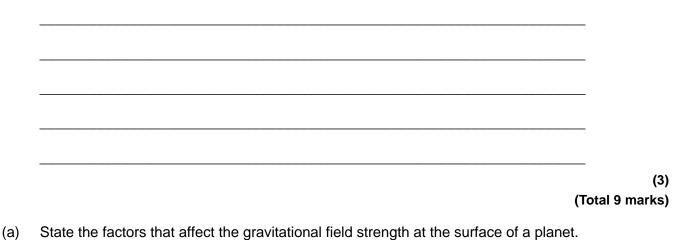
23



(a) (i) Using data from the graph, find the gravitational force on the lunar module and hence find its speed when its orbital height is 6.0×10^5 m.

(3)

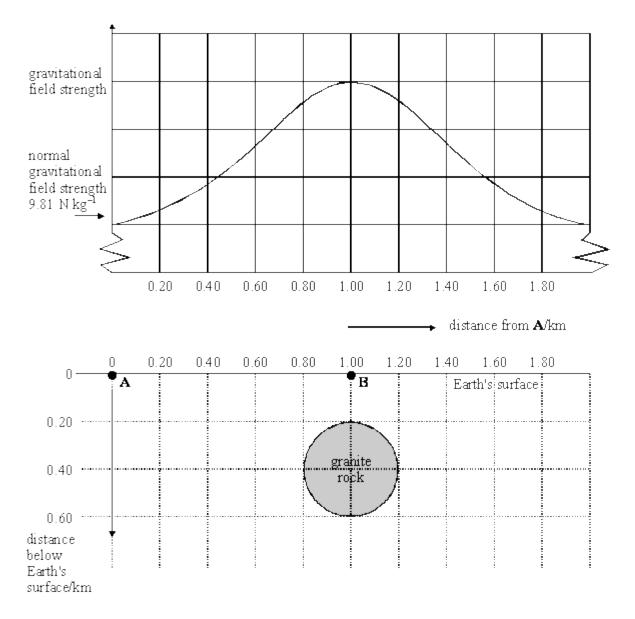
- Using data from the graph, find the change in gravitational potential energy of the (ii) lunar module as it descends from its orbit to the surface of the Moon.
- The descent of the lunar module is controlled by a set of rockets. Describe how you would (b) use the data which you have already calculated to determine the minimum fuel load which would enable the lunar module to land on the surface of the Moon and subsequently to rejoin its parent craft in orbit. State what additional information you would need to know.



State the factors that affect the gravitational field strength at the surface of a planet.

(3)

(b) The diagram below shows the variation, called an anomaly, of gravitational field strength at the Earth's surface in a region where there is a large spherical granite rock buried in the Earth's crust.



The density of the granite rock is 3700 kg m⁻³ and the mean density of the surrounding material is 2200 kg m⁻³.

(i) Show that the difference between the mass of the granite rock and the mass of an equivalent volume of the surrounding material is 5.0×10^{10} kg.

(ii) The universal gravitational constant $G = 6.7 \times 10^{-11}$ N m² kg⁻². Calculate the difference between the gravitational field strength at **B** and that at point **A** on the Earth's surface that is a long way from the granite rock.

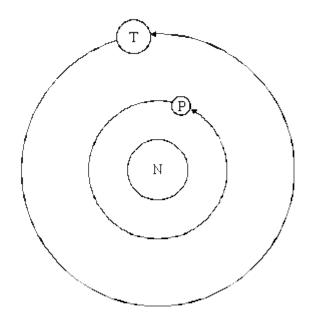
- (4)
- (iii) Add to the diagram above a graph to show how the variation in gravitational field strength would change if the granite rock were buried deeper in the Earth's crust.

(1)

(Total 11 marks)

25

The diagram below (not to scale) shows the planet Neptune (N) with its two largest moons, Triton (T) and Proteus (P). Triton has an orbital radius of 3.55×10^8 m and that of Proteus is 1.18×10^8 m. The orbits are assumed to be circular.



(a) Explain why the velocity of each moon varies whilst its orbital speed remains constant.

(1)

(b) Write down an equation that shows how Neptune's gravitational attraction provides the centripetal force required to hold Triton in its orbit. Hence show that it is unnecessary to know the mass of Triton in order to find its angular speed.

(c) Show that the orbital period of Triton the orbital period of Proteus is approximately 5.2.

(4) (Total 8 marks)

(3)

Mark schemes

(a)

1

2

	N kg⁻¹	electric field strength	N C ^{-1} or V m ^{-1}	(1)
gravitational constant	N m ² kg ⁻²			(1)
mass	kg	charge	С	(1)
distance (from mass to point)	m	distance (from charge to point)	m	(1)

(b) (i) none (1)

both
$$F_E$$
 and $F_G \propto \frac{1}{r^2}$ (hence both reduced to $\frac{1}{4}$ [affected equally] (1)

(ii) charge on B must be doubled (1)

(3)

[7]

(4)

(a) force of attraction between two point masses (or particles) (1)

proportional to product of masses (1)

inversely proportional to square of distance between them (1)

[alternatively

quoting an equation, $F = \frac{GM_1M_2}{r^2}$ with all terms defined (1)

reference to point masses (or particles) or *r* is distance between centres (1)

F identified as an attractive force (1)]

max 2

	(b)	(i)	mass of larger sphere $M_{\rm L}$ (= $\frac{4}{3}\pi r^3 \rho$) = $\frac{4}{3}\pi \times (0.100)^3 \times 11.3 \times 10^3$ (1)		
			= 47(.3) (kg) (1)		
			[alternatively		
			use of $M \mu r^3$ gives $\frac{M_L}{0.74} = \left(\frac{100}{25}\right)^3$ (1) (= 64)		
			and <i>M</i> _L = 64 × 0.74 = 47(.4) (kg) (1)]	2	
		(ii)	gravitational force F $\left(=\frac{GM_{L}M_{S}}{x^{2}}\right) = \frac{6.67 \times 10^{-11} \times 47.3 \times 0.74}{0.125^{2}}$ (1)	2	
			= 1.5 × 10 ⁻⁷ (N) (1)	2	
	(c)	for th	he spheres, mass μ volume (or μ r^3 , or $M = \frac{4}{3}\pi r^3 \rho$) (1)		
		mas	s of either sphere would be 8 × greater (378 kg, 5.91 kg) (1)		
		this	would make the force 64 × greater (1)		
		but s	separation would be doubled causing force to be 4 × smaller (1)		
			effect would be to make the force $(64/4) = 16 \times \text{greater}(1)$		
		(ie z	2.38 × 10 ^{−6} N)	max 4	[10]
3	(a)		that both astronaut and vehicle are travelling at same (orbital) speed or hav tripetal) acceleration / are in freefall	e the same	[10]
			Not falling at the same speed		
				B1	
		No (normal) reaction (between astronaut and vehicle)		
				B1	

(b) (i) Equates centripetal force with gravitational force using appropriate formulae

E.g.
$$\frac{GMm}{r^2} = \frac{mv^2}{r}$$
 or $mr\omega^2$ B1

Correct substitution seen e.g.
$$v^2 = \frac{6.67 \times 10^{-11} \times 5.97 \times 10^{24}}{\text{any value of radius}}$$

Β1

(Radius of) 7.28 × 10^6 seen or 6.38 × 10^6 + 0.9 × 10^6

B1

7396 (m s⁻¹) to at least 4 sf Or $v^2 = 5.47 \times 10^7$ seen

(ii)	ΔΡΕ = 6.67 × 10 ⁻¹¹ × 5.97 × 10 ²⁴ × 1.68 × 10 ⁴ (1 / (7.28 × 10 ⁶) - 1 / (6.78 × 10 ⁶))		
		C1	
	−6.8 × 10 ¹⁰ J		
		C1	
	ΔKE =0.5 × 1.68 × 10 ⁴ ×(7700 ² -7400 ²) = 3.81 × 10 ¹⁰ J		
		C1	
	$AKE = ABE = (-) 2.00 \times 10^{10} (1)$		
	$\Delta KE - \Delta PE = (-) 2.99 \times 10^{10} (J)$		
		A1	
	OR		
	Total energy in original orbit shown to be $(-)GMm/2r$ or $mv^2/2 - GMm/r$		
		C1	
	Initial energy = - 6.67 × 10 ⁻¹¹ × 5.97 × 10 ²⁴ × 1.68 × 10 ⁴ / (2 × 7.28 × 10 ⁶) = 4.59 × 10 ¹¹		
		C1	
	Final energy = - 6.67 × 10 ⁻¹¹ × 5.97 × 10 ²⁴ × 1.68 × 10 ⁴ / (2 × 6.78 × 10 ⁶) = 4.93 × 10 ¹¹		
	3.4×10^{10} (J)		
	Condone power of 10 error for C marks		
		A1	
		4	[10]
prop	active force between point masses (1) portional to (product of) the masses (1)		-
inve	rsely proportional to square of separation/distance apart (1)	3	

(a)

(b)
$$m\omega^2 R = (-)\frac{GMm}{R^2} \left(\text{or} = \frac{m\nu^2}{R} \right)$$
 (1)

(use of
$$T = \frac{2\pi}{\omega}$$
 gives) $\frac{4\pi^2}{T^2} = \frac{GM}{R^3}$ (1)

G and M are constants, hence $T^2 \propto R^3$ (1)

(c) (i) (use of
$$T^2 \propto R^3$$
 gives) $\frac{365^2}{(1.50 \times 10^{11})^3} = \frac{T_m^2}{(5.79 \times 10^{10})^3}$ (1)

 $T_{\rm m} = 87(.5) \text{ days (1)}$

(ii)
$$\frac{1^2}{(1.50 \times 10^{11})^3} = \frac{165^2}{R_N^3}$$
 (1) (gives $R_N = 4.52 \times 10^{12}$ m)

ratio =
$$\frac{4.51 \times 10^{12}}{1.50 \times 10^{11}}$$
 = 30(.1) (1)

[10]

4

3

(a) (i) $M = \frac{4}{3} \pi R^3 \rho \checkmark$

combined with $g_s = \frac{GM}{R^2}$ (gives $g_s = \frac{4}{3}\pi GR\rho$) \checkmark Do not allow r instead of R in final answer but condone in early

stages of working.

Evidence of combination, eg cancelling R^2 required for second mark.

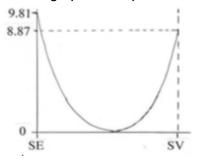
(ii)
$$R = \left(\frac{3g_s}{4\pi G\rho}\right) = \frac{3 \times 8.87}{4\pi 6.67 \times 10^{-11} \times 5.24 \times 10^3} \checkmark$$

gives $R = 6.06 \times 10^6$ (m) \checkmark answer to **3SF** \checkmark

SF mark is independent but may only be awarded after some working is presented.

(b) line starts at 9.81 and ends at 8.87 \checkmark

correct shape curve which falls and rises \checkmark falls to zeo value near centre of and to right of centre of distance scale \checkmark [*Minimum of graph in 3rd point to be >0.5 and <0.75 SE-SV distance*]



For 3rd mark accept flatter curve than the above in central region.

					r - 1
6	(a)	(i)	g gravitational field strength, G gravitational constant		
				C1	
			g force on 1 kg (on or close to) Earth's surface		
				A1	
			<i>G</i> universal constant relating attraction of any two masses to their separation/constant in Newton's law of gravitation		
				A1	3
		(ii)	equates w and cancels m		
				B1	1
		(iii)	substitutes values into equation		-
				B1	
			correct calculation 5.99 × 10^{24}		
				C1	
			answer to two significant figures 6.0 \times 10 ²⁴ (kg)		
				A1	3
	(b)	(i)	1 day/24 hours/86400 (s)		
				B1	
					1

3

[8]

(i)
$$4.24 \times 10^{7}$$
 (m)
B1
(ii) $v = 2\pi d$ T or equivalent
(iii) $conversion of period to seconds (allow in (b)(i))
(1)
3.08 (cao)
A1
3
(iv) communication/specific example of communication (eg
satellite TV/weather)
B1
1
(v) avoids dish having to track/stationary footprint
B1
1
(v) avoids dish having to track/stationary footprint
B1
1
(v) avoids dish having to track/stationary footprint
B1
1
(v) avoids dish having to track/stationary footprint
B1
1
(v) avoids dish having to track/stationary footprint
B1
1
(v) avoids dish having to track/stationary footprint
B1
1
(v) avoids dish having to track/stationary footprint
B1
4
(v) avoids dish having to track/stationary footprint
B1
4
(v) avoids dish having to track/stationary footprint
B1
1
(v) avoids dish having to track/stationary footprint
B1
1
(v) avoids dish having to track/stationary footprint
B1
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(v) avoids dish having to track/stationary footprint
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(v) avoids dish having to track/stationary footprint
B1
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(v) avoids dish having to track/stationary footprint
B1
1
(v) avoids dish having to track/stationary footprint
B1
1
(v) avoids dish having to track/stationary footprint
4
(b) Chemical combustion of propellant / fuel or gases produced at high pressure
Gas is expelled / expands through nozzle
Change in momentum of gases escaping
equal and opposite change in momentum
Na x 3
N3 in terms of forces worth 1$

(c) $0.031(4) \text{ (m s}^{-2})$

(d) Use of rocket equation

$$v = 1200 \ln \frac{3050}{1330}$$

996 (m s⁻¹)
Condone 1000 (m s⁻¹)

3

3

3

1

(e) (i) Use of correct mass 108 kg

$$F = \frac{6.67 \times 10^{-11} \times 1.1 \times 10^{13} \times 108}{(2 \times 10^3)^2}$$

0.0198 N

Allow incorrect powers of 10 and mass

(ii) Use of
$$v = \sqrt{\frac{2GM}{r}}$$

Correct substitution $v = \frac{2 \times 6.67 \times 10^{-11} \times 1.1 \times 10^{18}}{2 \times 10^8}$

0.86 (m s⁻¹)

(iii) Impulse = 25 N × 4.8 = 120 N s

$$(120 = 108 v so)$$
 Velocity = 1.1 m s⁻¹

Clear conclusion

8

ie explanation/comparison of calculated velocity with escape velocity from **(e)(ii)**

May use F = ma approach

[20]

3

(a) (i) (Minimum) Speed (given at the Earth's surface) that will allow an object to leave / escape the (Earth's) gravitational field (with no further energy input)
 Not gravity
 Condone gravitational pull / attraction

B1

(ii)
$$\frac{1}{2}mv^2 = \frac{GMm}{r}$$

			B1		
		Evidence of correct manipulation At least one other step before answer			
			B1	2	
	(iii)	Substitutes data and obtains $M = 7.33 \times 10^{22}$ (kg) or			
		Volume = $(1.33 \times 3.14 \times (1.74 \times 10^6)^3$ or 2.2×10^{19}			
		$or \rho = \frac{3v^2}{8\pi Gr^2}$			
			C1		
		3300 (kg m ⁻³)			
			A1	2	
(b)	adde	given all their KE at Earth's surface) energy continually ed in flight / continuous thrust provided / can use fuel tinuously)			
			B1		
		s energy needed to achieve orbit than to escape from h's gravitational field / it is not leaving the gravitational			
			B1	2	
				2	[7]
(a)	(i)	force per unit mass \checkmark a vector quantity \checkmark			
		Accept force on 1 kg (or a unit mass).		2	

(ii) force on body of mass *m* is given by $F = \frac{GMm}{(R+h)^2} \checkmark$

gravitational field strength
$$g\left(=\frac{F}{m}\right)=\frac{GM}{(R+h)^2}$$

For both marks to be awarded, correct symbols must be used for M and m.

(b) (i)
$$F\left(=\frac{GMm}{(R+h)^2}\right) = \frac{6.67 \times 10^{-11} \times 5.98 \times 10^{24} \times 2520}{\left(\left(6.37 \times 10^6\right) + \left(1.39 \times 10^7\right)\right)^2} \checkmark$$

$$= 2.45 \times 10^3$$
 (N) \checkmark to **3SF** \checkmark

 1^{st} mark: all substituted numbers must be to at least 3SF. If 1.39×10^7 is used as the complete denominator, treat as AE with ECF available.

3rd mark: SF mark is independent.

(ii)
$$F = m\omega^2 (R + h)$$
 gives $\omega^2 = \frac{2450}{2520 \times 2.03 \times 10^7} \checkmark$

from which ω = 2.19 × 10⁻⁴ (rad s⁻¹) \checkmark

time period
$$T\left(=\frac{2\pi}{\omega}\right) = \frac{2\pi}{2.19 \times 10^{-4}}$$
 or $= 2.87 \checkmark 10^4 \text{ s} \checkmark$

$$[\text{or } F = \frac{mv^2}{R+h} \text{ gives } v^2 = \frac{2.45 \times 10^3 \times ((6.37 \times 10^6) + (13.9 \times 10^6))}{2520} \checkmark$$

from which $v = 4.40 \checkmark 10^3$ (m s⁻¹) \checkmark

time period
$$T\left(=\frac{2\pi(R+h)}{v}\right) = \frac{2\pi \times 2.03 \times 10^7}{4.40 \times 10^3}$$
 or $= 2.87 \times 10^4$ s \checkmark]

$$[or \ T^2 = \frac{4\pi^2 (R+h)^3}{GM} \checkmark$$

$$=\frac{4\pi^2\left((6.37\times10^6)+(13.9\times10^6)\right)^3}{6.67\times10^{-11}\times5.98\times10^{24}}\checkmark$$

gives time period T = $2.87 \times 10^4 \text{s} \checkmark$]

$$=\frac{2.87\times10^4}{3600}=7.97 \text{ (hours) } \checkmark$$

number of transits in 1 day = $\frac{24}{7.97}$ = 3.01 (\approx 3) \checkmark

Allow ECF from wrong F value in (i) but mark to max 4 (because final answer won't agree with value to be shown).

First 3 marks are for determining time period (or frequency). Last 2 marks are for relating this to the number of transits.

Determination of $f = 3.46 \times 10^{-5} (s^{-1})$ is equivalent to finding T by any of the methods.

(c) acceptable use
$$\checkmark$$

satisfactory explanation √

- e.g. monitoring weather or surveillance:
 - whole Earth may be scanned or Earth rotates under orbit
 - or information can be updated regularly
- or communications: limited by intermittent contact
- or gps: several satellites needed to fix position on Earth

Any reference to equatorial satellite should be awarded 0 marks.

[14]

2

		B1	
	energy input needed to move to infinity (from the point) work done by the field moving object from infinity potential energy falls as object moves from infinity		
		B1	2
(b)	Any pair of coordinates read correctly		
	±1/2 square	C1	
	Use of $E_{\rho \text{ or }} V = (-) \frac{GM}{r}$		
	Rearrange for M	C1	
	$6.4 (\pm 0.5) \times 10^{23} \text{ kg}$		
		A1	3
(c)	Reads correct potential at surface of Mars = -12.6 (MJ)		
		C1	
	or reads radius of mars correctly (3.5×10^6)		
	equates to $\frac{1}{2}$ v ² (condone power of 10 in MJ)		
	use of v = $\sqrt{2GM/r}$ with wrong radius	C1	
	5000 ± 20 m s ⁻¹ (condone 1sf e.g. 5 km s ⁻¹)		
		A1	
	e.c.f. value of M from (b) may be outside range for other me	thod 6.2	

× 10⁻⁹x √their M

			B1		
		Many values give 4.2 so allow mark is for reading and usi correct coordinates but allow minor differences in readings Ignore powers of 10 but consistent	ng		
		Two correct calculation of Vr			
			B1		
		Three correct calculations with conclusion			
			B1		
				3	[11]
11	(a)	mass depends only on the amount of matter present owtte			
			B1		
		weight is force between body and Earth/depends on <i>g/mg/</i> gravitational field strength or answers in terms of Newton's gravitational law			
			B1		
		g (etc) varies at different points on and above the Earth or is different on different planets etc			
			B1	3	

		B1	
	energy has to be put in/work has to be done to move mass to infinity or a bodies energy/PE decreases as a body moves from infinity towards the Earth		
		B1	2
(ii)	need to show Vr to be constant, clear from algebra or final statement		
		B1	
	two sets of data used correctly		
		B1	
	all three sets of data used correctly (4.02, 4.025, 4.028)	B1	
			3
(iii)	energy change per kg = $(5.36 - 3.22) \times 10^7$ (J)	D4	
	total change = 963 (960) × 10^7 J	B1	
	101ai change - 303 (300) × 10 3	B1	
		2.	2

C1

$$v^2 = 3.2 \times 10^7 \text{m}^2 \text{s}^{-2} \text{ or } v = 5670 \text{ ms}^{-1}$$

Use of KE = $\frac{1}{2} mv^2$ using their v
C1
7.2 GJ
(ii) KE changes by 4.8 GJ (allow ecf, 12 – their ci)
B1
(iii) total energy (supplied) = (4.8) GJ (cnao)
(allow 5.2 GJ using 10 GJ for change in E_p)
(allow variations due to rounding off if physics
is correct in previous parts)
B1
1

12 (a)
$$\omega \left(=\frac{2\pi}{T}\right) = \frac{2\pi}{97 \times 60}$$
 [or $\omega \left(=\frac{360}{T}\right) = \frac{360}{97 \times 60}$]
 $= 1.1 \times 10^{-3} (1.08 \times 10^{-3}) (1) [= 6.2 (6.19) \times 10^{-2}]$
rad s⁻¹ [accept s⁻¹] (1) [degree s⁻¹]
(b) (i) $\frac{GMn}{r^2} = m \omega^2 r$ or $r^3 = \frac{GM}{\omega^2}$ (1)
gives $r^3 = \frac{6.67 \times 10^{-11} \times 5.98 \times 10^{24}}{(1.08 \times 10^{-3})^2}$ (1)
 $\therefore r = 6.99 \times 10^6$ (m) (1)

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[16]

3

(ii)
$$F(=m\omega^2 t) = 1.1 \times 10^4 \times (1.08 \times 10^{-3})^2 \times 6.99 \times 10^6 (1)$$

= 9.0 × 10⁴ (8.97 × 10⁴) (N) (1)
[or $F\left(=\frac{Gh/m}{r^2}\right) = \frac{6.67 \times 10^{-11} \times 5.98 \times 10^{24} \times 1.1 \times 10^4}{(6.99 \times 10^6)^2}$ (1)
= 9.0 × 10⁴ (8.98 × 10⁴) (N) (1)]
2
[8]
(a) (i) relationship between them is $E_p = mV$ (allow $\Delta E_p = m\Delta V$) [or V
is energy per unit mass (or per kg)] (1)
(ii) value of V_p is doubled (1)
value of V is unchanged (1)
2
(b) (i) use of $V = -\frac{Gh/t}{r}$ gives $r_A = \frac{6.67 \times 10^{-11} \times 5.98 \times 10^{24}}{12.0 \times 10^6}$ (1)
= 3.3(2) × 10⁷ (m) (1)
2
(ii) since $V \propto (-)\frac{1}{r} \left(cr \frac{r_A}{r_B} = \frac{v_B}{v_A} = \frac{38.0}{12.0} = 3 \right) r_B = \frac{3.32 \times 10^7 \text{ m}}{3}$ (1)
(which is = 1.1 × 10⁴ km)
1
(iii) centripetal acceleration $g_B = \frac{Gh/t}{r_B^2} = \frac{6.67 \times 10^{-11} \times 5.98 \times 10^{24}}{(1.11 \times 10^7)^2}$ (1)
[alternatively, since $g_B = (-)\frac{v_B}{v_A}, g_B = \frac{36.0 \times 10^6}{1.11 \times 10^7}$ (1)
= 3.2 (m s⁻²) (1]
2

(iv) use of $\Delta E_p = m\Delta V$ gives $\Delta E_p = 330 \times (-12.0 - (-36.0)) \times 10^6$ (1) (which is 7.9 × 10⁹ J or ≈ 8 GJ)

(c) g is not constant over the distance involved

(or *g* decreases as height increases or work done per metre decreases as height increases or field is radial and/or not uniform) (1)

[10]

1

2

3

(a) work done/energy change (against the field) per unit mass (1) when moved from infinity to the point (1)

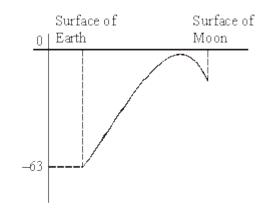
(b)
$$V_{\rm E} = -\frac{GM_{\rm E}}{R_{\rm E}}$$
 and $V_{\rm M} = -\frac{GM_{\rm M}}{R_{\rm M}}$ (1)

$$V_{\rm M} = -G \times \frac{M_{\rm E}}{81} \times \frac{3.7}{R_{\rm E}} = \frac{3.7}{81} V_{\rm E}$$
 (1)

=
$$4.57 \times 10^{-2} \times (-63) = -2.9 \text{ MJ kg}^{-1}$$
 (1) (2.88 MJ kg⁻¹)

(c)

14



limiting values $(-63, -V_M)$ on correctly curving line (1) rises to value close to but below zero (1) falls to Moon (1) from point much closer to M than E (1)

max 3

(a)

(i)
$$h (= ct) (= 3.0 \times 10^8 \times 68 \times 10^{-3}) = 2.0(4) \times 10^7 \text{ m} (1)$$

 $g = (-) \frac{GM}{r^2}$ (1)

 $r (= 6.4 \times 10^{6} + 2.04 \times 10^{7}) = 2.68 \times 10^{7} \text{ (m)}$ (m) (1) (allow C.E. for value of *h* from (i) for first two marks, but not 3rd)

$$g = \frac{6.67 \times 10^{-11} \times 6.0 \times 10^{24}}{(2.68 \times 10^7)^2}$$
(1) (= 0.56 N kg⁻¹)

4

[8]

(b) (i)
$$g = \frac{v^2}{r}$$
 (1)
 $v = [0.56 \times (2.68 \times 10^7)]^{\frac{1}{2}}$ (1)
 $= 3.9 \times 10^3 \text{m s}^{-1}$ (1) $(3.87 \times 10^3 \text{ m s}^{-1})$
(allow C.E. for value of *r* from a(ii)
[or $v^2 = \frac{GM}{r} =$ (1)
 $v = \left(\frac{6.67 \times 10^{-11} \times 6 \times 10^{24}}{2.68 \times 10^7}\right)^{\frac{1}{2}}$ (1)

(ii)
$$T\left(=\frac{2\pi r}{v}\right) = \frac{2\pi \times 2.68 \times 10^7}{3.87 \times 10^3}$$
 (1)

= $4.3(5) \times 10^4$ s (1) (12.(1) hours) (use of $v = 3.9 \times 10^3$ gives $T = 4.3(1) \times 10^4$ s = 12.0 hours) (allow C.E. for value of v from (I)

[alternative for (b):

(a)

16

(i)
$$v\left(\frac{2\pi r}{T}\right) = \frac{2\pi \times 2.68 \times 10^7}{4.36 \times 10^4}$$
 (1)

= 3.8(6) × 10³ m s⁻¹ (1)]

(allow C.E. for value of r from (a)(ii) and value of T)

(ii)
$$T^2 = \left(\frac{4\pi^2}{GM}\right) r^3$$
 (1)
 $\left(=\frac{4\pi^2}{6.67 \times 10^{-11} \times 6.0 \times 10^{24}} \times (2.68 \times 10^7)^3\right) = (1.90 \times 10^9 \text{ (s}^2) \text{ (1)}$
 $T = 4.3(6) \times 10^4 \text{ s (1)}$

max 2

5

[9]

(b) (i)
$$\frac{GMm}{r^2} = m\omega^2 r$$
 (1)
$$T = \frac{2\pi}{\omega}$$
 (1)

$$r \left(= \frac{GMT^2}{4\pi^2} \right)^{1/3} = \left(\frac{6.67 \times 10^{-11} \times 6 \times 10^{24} \times (24 \times 3600)^2}{4\pi^2} \right)^{1/3}$$
(1)

(gives $r = 42.3 \times 10^{3}$ km)

(ii)
$$\Delta V = GM\left(\frac{1}{R} - \frac{1}{r}\right)$$
 (1)
 $= 6.67 \times 10^{-11} \times 6 \times 10^{24} \times \left(\frac{1}{6.4 \times 10^6} - \frac{1}{4.23 \times 10^7}\right)$
 $= 5.31 \times 10^7 \text{ (J kg}^{-1)}$ (1)
 $\Delta E_p = m\Delta V (= 750 \times 5.31 \times 10^7) = 3.98 \times 10^{10} \text{ J}$ (1)
(allow C.E. for value of ΔV)
[alternatives:

calculation of
$$\frac{GM}{R}$$
 (6.25 × 10⁷) or $\frac{GM}{r}$ (9.46 × 10⁶) (1)

or calculation of $\frac{GMm}{R}$ (4.69 × 10¹⁰) or $\frac{GMm}{r}$ (7.10× 10⁹) (1) calculation of both potential energy values (1)

subtraction of values or use of $m\Delta V$ with correct answer (1)]

(a) (i)

17

(ii) increase in potential energy =
$$m\Delta V$$
 (1)
= 1200 × (62 - 21) × 10⁶ (1)
= 4.9 × 10¹⁰ J (1)

(b) (i) $g = -\frac{\Delta V}{\Delta x}$ (1)

(4)

(ii) g is the gradient of the graph =
$$\frac{62.5 \times 10^6}{4 \times 6.4 \times 10^6}$$
 (1)
= 2.44 N kg⁻¹ (1)

(iii) $g \propto \frac{1}{R^2}$ and *R* is doubled (1)

expect g to be $\frac{9.81}{4} = 2.45 \text{ N kg}^{-1}$ (1)

[alternative (iii)

$$g \propto \frac{1}{R^2}$$
 and *R* is halved (1)

expect g to be 2.44 x 4 = 9.76 N kg⁻¹ (1)]

(5)

[9]

(a) attractive force between two particles (or point masses) (1) proportional to product of masses and inversely proportional to square of separation [or distance] (1)

(b) (for mass, *m*, at Earth's surface)
$$mg = \frac{GMm}{R^2}$$
 (1) rearrangement gives result (1)

(c) $M_{\text{moon}}\left(=\frac{gR^2}{G}\right) = \frac{1.62 \times (1.74 \times 10^6)^2}{6.62 \times 10^{24}}$ (1)

= 7.35 × 10²² kg (1)

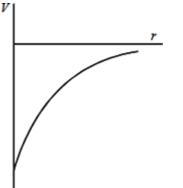
$$\frac{M_{\text{moon}}}{M_{\text{earth}}} = \frac{7.35 \times 10^{22}}{6.00 \times 10^{24}} \ (= 0.0123) \ \therefore \ 1.23\%$$

2

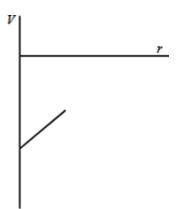
2

3

[7]



gradient decreases as r increases (1) V increases as r increases (1) only negative values of V shown (1)



constant gradient (1) V increases as r increases (1)

20

(a) (i) force per unit mass/force per kg

(ii) N kg⁻¹ **not** ms⁻² alone

[max 4]

2

Β1

B1

(b) (i) *GM/R*² seen

21

				C1			
			$GM_Q/(3R)^2$ seen				
				C1			
			mass of $Q = 9M$				
		(::)	nearcosthrough (2.D. c) and falls off in surve	A1			
		(ii)	passes through (3 <i>R</i> , <i>g</i>) and falls off in curve	M1			
			two further points checked e.g., $(6R,g/4)$ (12R, g/16)				
				M1			
			overall line quality – single smooth line (both Ms for this)				
				A1	6		
1							I
	(a)	force	e acting per unit mass or $g = F / m$ or $g = \frac{GM}{R^2}$ with terms defined			(1)	
	(b)	(i)	direction of F_{E} correct in each diagram		B1		
			direction of $F_{\rm M}$ correct in each diagram		B1		
			direction of F_{S} correct in each diagram		DI		
					B1		
			$F_{\rm S}$ must be distinguished from $F_{\rm M}$				
			penalty of 1 mark for any missing labelling			(3)	
		(ii)	sun and moon pulling in same direction / resultant of $F_{\rm M}$ and $F_{\rm S}$ is clear response including summation of $F_{\rm M}$ and $F_{\rm S}$	greatest /			
					M1		
			configuration A		A1	(2)	
						(=)	

[8]

(c) $F = GMm / R^2$

correct substitution
$$\frac{6.7 \times 10^{-11} \times 2.0 \times 10^{30}}{(1.5 \times 10^{11})^2}$$
C1

(5.95 or 5.96 or 5.9 or 6.0)
$$\times 10^{-3} \mbox{ N kg}^{-1}$$

A1

(3)

C1

(a) (i) force per unit mass (allow equation with defined terms) 22 **B1** (1) (ii) diagram of method that will work (pendulum / light gates / solenoid and mechanical gate / strobe photography / video) **B1** pair of measurements (eg length of pendulum and (periodic) time / distance and time of fall - could be shown on diagram) **M1** instruments to measure named quantities (may be on diagram) A1 correct procedure (eg calculate period for range of lengths, measure the time of fall for range of heights) **B1** good practice - series of values and averages / use of gradient of graph **B1** appropriate formula and how g calculated **B1** (6) evidence of gr^2 being used (b) (i) **C1** values of 0.25, 0.11, 0.06(25) no s.f. penalty here unless values given as fractions A1 (2) (ii) points correctly plotted on grid (e.c.f.) **B1** smooth curve of high quality at least to 10×10^7 m, no intercept on r axis **B1** (2) (iii) attempt to use area under curve

	B1
evidence of × 800 kg	

$$(4.3 - 5.3) \times 10^9 \text{ J}$$

use of equation for potential $\Delta E_G = m(g_1r_1 - g_2r_2)$

B1 evidence of × 800 kg

$$(4.7 - 4.9) \times 10^9$$
 J

```
max 2 if assumed values of G and M used
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allow calculation of *GM* from graph followed by substitution into $\Delta E_G = M_G(m / r_1 - m / r_2)$ for 3 marks

[14]

(3)

B1

B1

B1

23

(a)

(i)

F = 2500 or 2600 N

 $F = mv^2 / r$

1500 m s⁻¹ / 1480 m s⁻¹ or any further progress towards a solution such as attempting to use: $F = GMm / r^2$ or $\frac{1}{2} mv^2 = mgh$ or equations of motion or $r = 6 \times 10^5$ or $F = mv^2 / (r + h)$

or evidence of time wasted, for example, repeated attempts at a solution no unit penalty or s.f. penalty

(3)

B1

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	(ii)	using the area under the graph between 0 ar	nd 6.0 × 10 ⁵ m				
		or use of $V_G = GM / r$ or $PE = GMm / r$			C1		
		between 20 and 22 squares					
		or 1 square = 10^8 J					
		or uses the trapezium rule or rectangle and (allow if they fail with powers of ten)	triangle				
		or attempts to find the difference between th	ne 2 values of PE		C1		
		2.0×10^9 J to 2.2×10^9 J					
		or attempt to complete the calculation (may be confounded by lack of <i>r</i>)					
					A1	(3)	
(b)	(wor	k done by motors) relates to change in PE or	(PE + KE)		B 1		
	2 × (PE + KE) condone 2 × (PE)			B1		
		need to know energy value of fuel / fuel density / energy density of fuel / fuel economy / efficiency of engines			DI		
					B1	(3)	[9]
(a)	the r	adius/diameter of the planet no	ot 'size'				
				B1			
	the r	nass (or density) of the planet					
				B1			

(b)	(i)	volume of the granite = $4/3\pi r^3$ or radius of the granite = 0.2 km (may be seen in an incorrect equation)	
			B1
		200^3 or $4/3\pi 0.2^3$ or 3.35×10^7 m ³	
			B1
		Mass = density × volume used with any density and their volume (Volume may be in formula form) If they use correct volume then either 1.24×10^{11} or 7.37×10^{10} gets the mark)	
			B1
		$(3700-2200) \times 3.35 \times 10^7$ or $1500 \times 3.35 \times 10^7$ kg or $(1.24 \times 10^{11} - 7.37 \times 10^{10})$ or 5.025×10^{10} or 5.03×10^{10} seen Condone rounding off early leading to 4.6×10^{10} kg	
			B1
		NB	
		4) the fourth meric is not for $E \cap (10^{10})$ all working much be	

1) the fourth mark is not for 5.0×10^{10} – all working must be shown

2) those who do not show conversion of radius from km to m in the calculation but otherwise correct will get 3

(ii)	Gravitation field strength $g = GM/r^2$ or		
	uses distance of 0.4 km for r		
		C1	
	Substitution for extra field strength = $6.7 \times 10^{-11} \times 5.0 \times 10^{10}/(0.4 \times 10^3)^2$ Condone <i>r</i> = 0.4 for this mark		
		C1	
	Correct substitution for the extra field strength with correct powers of 10		
		C1	
	$2.1 \times 10^{-5} \text{ N kg}^{-1}$ (condone m s ⁻²)		
	or 1.9×10^{-5} if 4.6 × 10 ¹⁰ carried forward from (i)		
		A1	4
(iii)	Correct general shape always below original curve		
		B1	1
Alte	ernative scheme for different approach to (ii)		
(ii)	Gravitation field strength = GM/r^2 or		
	uses distance of 0.4 km for <i>r</i>		
		C1	
	Correct substitution for field strength for granite (or soil) $6.7 \times 10^{-11} \times 1.24 \times 10^{11}/(0.4 \times 10^3)^2$ or $6.7 \times 10^{-11} \times 7.37 \times 10^{10}/(0.4 \times 10^3)^2$ Condone r = 0.4 for this mark		
		C1	
	Correct substitution for field strength for soil (or granite)		
		C1	
	$2.1 \times 10^{-5} \text{ N kg}^{-1}$ (condone m s ⁻²)		
		A1	4
			-

(a)

(b) Newton's law equation

centripetal force equation

cancel mass of Triton

(c)
$$\omega = 2\pi f \text{ or } \omega = 2\pi/T$$

$$\omega^2 r^3$$
 = constant or $\omega^2 = \frac{GM}{r^3}$

$\frac{T_T^2}{T_P^2} = \frac{r_T^3}{r_P^3} \text{ or statement of Kepler III for B3}$ $\frac{T_T}{T_P} = \sqrt{\frac{(3.55 \times 10^8)^3}{(1.18 \times 10^8)^3}} = 5.2(2)$

[8]

4

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Β1

M1

M1

A1

M1

M1

M1

1